

## Stability and Transition

A) Solution Techniques

B) Promotion / Suppression

C) Transition

Reading: Sch - 434-473.  
C & B - 290-301

A) Soln Techniques: Recall O-S eqn is 4<sup>th</sup> order complex ODE in  $y$ . We would like to solve for  $v(y)$ ,  $w$  or  $\omega$  for linear or spatial problem given mean flow:  $U_c, U(y)$ .

Rewrite O-S as a system of 1<sup>st</sup> order ODEs

$$(\alpha U - \omega)(v'' - \alpha^2 v) - \alpha U' v$$

$$+ \frac{i}{Re} [v'''' - 2\alpha^2 v'' + \alpha^4 v] = 0$$

$$\begin{aligned} (\phi) \quad u_1 &= v \\ (b) \quad u_2 &= v' \\ (s) \quad u_3 &= v'' - \alpha^2 v \\ (g) \quad u_4 &= u_3' \end{aligned}$$

$$\begin{aligned} (\alpha U - \omega)(u_3) - \alpha U' u_1 \\ + \frac{i}{Re} [u_4' + \alpha^4 u_1] \end{aligned}$$

=&gt;

~~$$u_1' - u_2 = 0$$~~

~~$$u_2' - \alpha^2 u_1 - u_3 = 0$$~~

~~$$u_3' - u_4 = 0$$~~

$$u_4' + \frac{i}{Re} [(\alpha U - \omega) u_3 - \alpha U' u_1] + \alpha^4 u_1 = 0$$

$$\Rightarrow \begin{aligned} u_1' - u_2 &= 0 \\ u_2' - \alpha^2 u_1 - u_3 &= 0 \\ u_3' - u_4 &= 0 \\ u_4' - \alpha^2 u_3 - i \operatorname{Re} [(\alpha U - \omega) u_3 - \alpha U^* u_1] &= 0 \end{aligned}$$

Boundary Conditions

$$y=0 : \quad u_1 = 0, \quad u_3 = 1$$

$$y = \eta \approx \delta : \quad u_4 + \xi u_3 = 0 \quad \xi^2 = \alpha^2 + i \operatorname{Re}(\alpha U - \omega)$$

$$u_2 + \frac{u_3}{\alpha + \xi} + \alpha u_1 = 0$$

consideration  
perturbations must  
vanish at  $y = \delta$

• Drop  $v(0) = 0$  for  $v''(0) = 1$

Discretize the above system using the trapezoidal rule to obtain a banded coefficient matrix -  $4 \times 4$  blocks with complex entries (2 equations) at each  $j$ . This system is non-linear ( $\alpha^2, \alpha^4, \dots$ ) so we use Newton-Raphson to solve

• Global variable -  $\alpha$

• Constraint - magnitude of  $v$  - example  $\int |v| dy = 1$  (normalization)

$$v''(0) = 1 \quad \text{and drop } v'(0) = 0$$

→ impose a non trivial solution, vary parameters in appropriate combinations to satisfy  $v'(0) = 0$

Spatial problem

given  $u(y/\theta, H)$

Choose  $R_c, \omega_r \rightarrow$  solve for  $\alpha$

$$\left. \begin{aligned} \alpha^*_{real}(R, \omega^*, H) &= \alpha_{real} \theta \\ \alpha^*_{imag}(R, \omega^*, H) &= \alpha_{imag} \theta \end{aligned} \right\} \text{non-dimensional}$$

↑  
 $u(y/\theta, H)$

$$R = \frac{u_c \theta}{\nu} = Re_\theta$$

$$\omega^* = \frac{\omega \theta}{u_c}$$

— x —

for is hand out

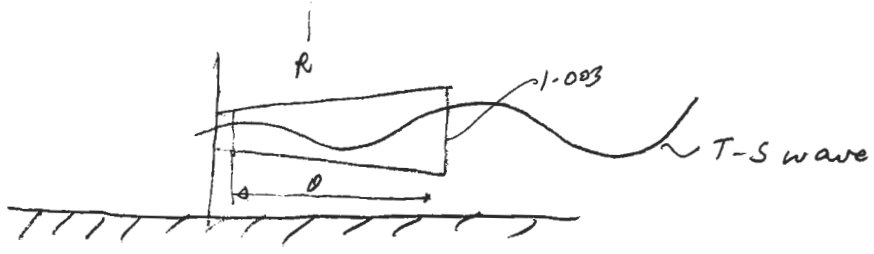
$$\vec{v}(y), u(y) = \frac{i}{\alpha} v(y)$$

- $\hat{u}_i$ : max at inflection point
- $H=8, \alpha_i < 0 \rightarrow$  large perturbations
- $H=2.5, \alpha_i > 0 \rightarrow$  decay of perturbations

for  $H=2.5$



$\Rightarrow$  wave grows 0.3% in 1  $\theta$  streamwise distance

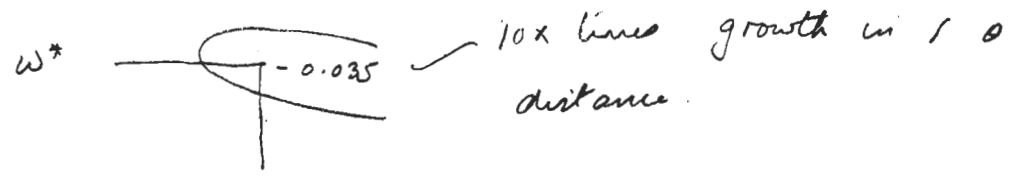


$$\lambda = \frac{2\pi}{\alpha_r} = \frac{2\pi\theta}{\alpha^*_r} \sim 63\theta$$

$$\sim 68$$

$$\theta \sim 8/10$$

For  $H = 3.2$



Briefly, note impact of suction, ht transfer, Mach # on stability (boundary)

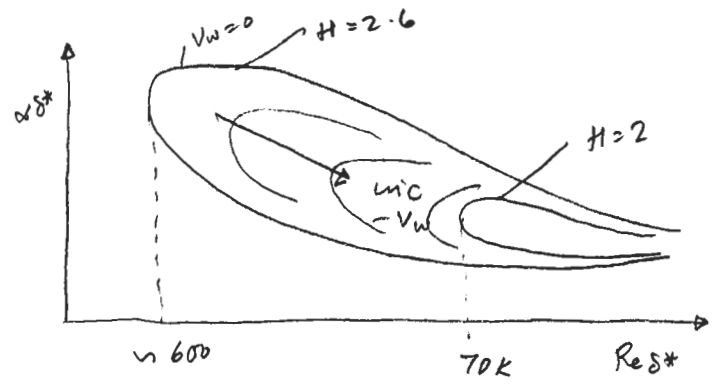
① Impact of suction:

Stability calculation using asymptotic suction profile

$$u/u_\infty(y) = 1 - e^{-\sqrt{v_w}y/\nu}, \quad v_w < 0$$

$$\delta^* = \frac{\nu}{-v_w}$$

and  $Re\delta^*_{crit} = 70,000$ .



Critical to minimize suction required to maintain stability

② HE transfer: To or from the surface can enhance or lower stability limit. HE transfer effects curvature of wall.

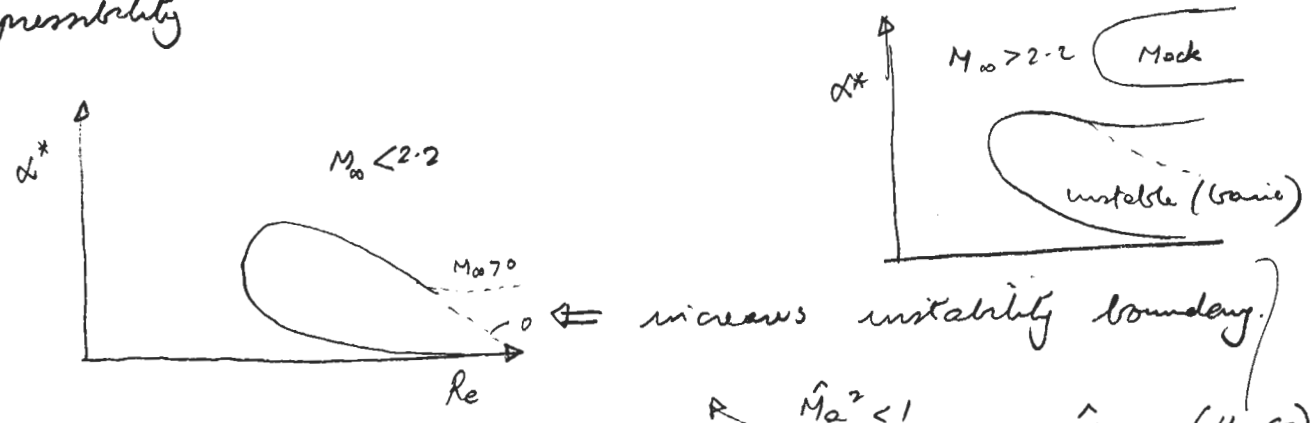
$$U''_w = -\frac{1}{\mu_w} \cdot \frac{\partial \mu}{\partial y} U'_w$$

If  $T_w > T_e$  (or  $T_\infty$ ), temperature is decreasing away from the wall  $\Rightarrow \frac{\partial T}{\partial y}|_w < 0$ , therefore  $\frac{\partial \mu}{\partial y}|_w < 0$  ( $\frac{d\mu}{dT} > 0$  for gas)

$\therefore U''_w > 0$  implying an inflection in profile

Opposite effect - stabilizing for cooling.

③ Compressibility



$C_{\alpha_r} = W$

$c_r$  - phase velocity

multiple modes are due change in character of equations - elliptical  $\rightarrow$  hyperbolic

local rel. M  $\hat{M}_a = \frac{U - c_r}{a_\infty}$   
 $\hat{M}_a^2 < 1$   
 $\hat{M}_a^2 > 1 @ M_\infty = 2.2$

extension of inflection point condition

$$\left. \frac{d}{dy} \left( \rho \frac{dU}{dy} \right) \right|_{y_s} = 0 \quad \text{at some } y_s \text{ in B-L.}$$

