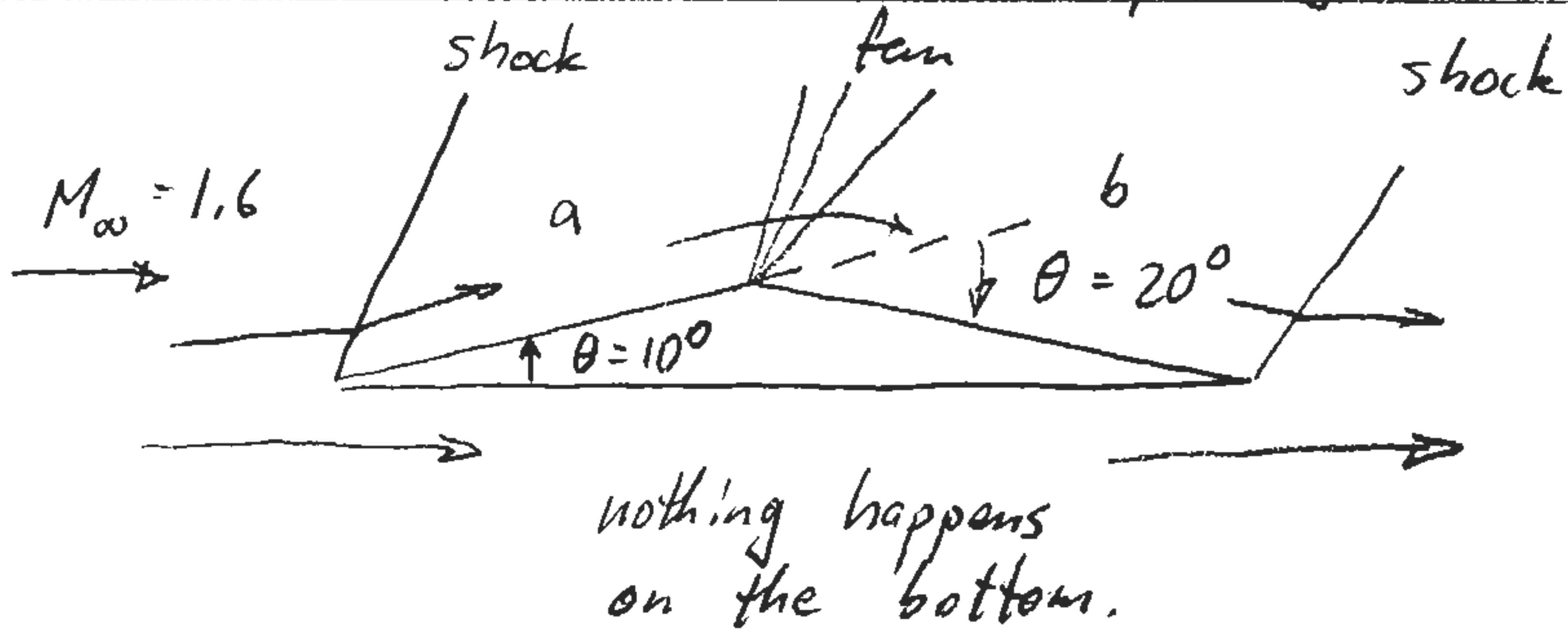


Flow sketch



a) Oblique Shock. $M_1 = 1.6, \theta = 10^\circ \rightarrow \beta = 51^\circ$ (chart, Anderson p. 513)

$$M_{n1} = M_1 \sin 51^\circ = 1.6 \cdot 0.777 = 1.243$$

$$p_2/p_1 = f(M_{n1}) = 1.636 \quad (\text{Anderson eq. 9.16, or Appendix B table.})$$

$$\text{Since } p_1 = p_\infty \rightarrow p_2 = \boxed{1.636 p_\infty = p_a}$$

$$M_{n2} = f(M_{n1}) = 0.817 \quad (\text{eq. 9.14})$$

$$\boxed{M_2 = M_a = \frac{M_{n2}}{\sin(51^\circ - 10^\circ)} = 1.245}$$

b) Expansion fan. $M_1 = M_a = 1.245, \nu(M_1) = 4.7^\circ$

$$\nu(M_2) = \nu(M_1) + 20^\circ = 24.7^\circ \rightarrow M_2 = 1.94$$

$$p_2 = p_{02} \left[1 + \frac{\gamma-1}{2} M_2^2 \right]^{-\frac{\gamma}{\gamma-1}}$$

$$p_{02} = p_{01} = p_{0a} \quad (\text{behind oblique shock.})$$

$$p_{0a}/p_{0\infty} = f(M_{n1}) = 0.987$$

$$p_{0a} = 0.987 p_{0\infty} = 0.987 p_\infty \left[1 + \frac{\gamma-1}{2} M_\infty^2 \right]^{\frac{\gamma}{\gamma-1}} = 4.195 p_\infty$$

$$\therefore p_2 = 4.195 p_\infty \left[1 + \frac{\gamma-1}{2} M_2^2 \right]^{-\frac{\gamma}{\gamma-1}} = \boxed{0.588 p_\infty = p_b}$$

Note: If we neglect the oblique shock's loss: $p_{0b} = p_{0a} = p_{0\infty} \Rightarrow p_2 = 0.596 p_\infty$ (not quite correct)

$$c) L' = L'_a + L'_b = (p_\infty - p_a) \frac{c}{2} + (p_\infty - p_b) \frac{c}{2} = (1 - 1.636) p_\infty \frac{c}{2} + (1 - 0.588) p_\infty \frac{c}{2}$$

$$\boxed{L' = -0.112 p_\infty c}$$

$$D' = D'_a + D'_b = -L'_a \tan 10^\circ + L'_b \tan 10^\circ = [-(1 - 1.636) + (1 - 0.588)] \tan 10^\circ p_\infty \frac{c}{2}$$

$$\boxed{D' = 0.0924 p_\infty c}$$

$$\text{Using } \frac{1}{2} \rho_\infty V_\infty^2 = \frac{\gamma}{2} p_\infty M_\infty^2 = 1.792 p_\infty \rightarrow$$

$$\boxed{C_L = L' / \frac{1}{2} \rho_\infty V_\infty^2 = -0.0625}$$

$$\boxed{C_D = D' / \frac{1}{2} \rho_\infty V_\infty^2 = 0.0516}$$